Item	Inspection/Maintenance Action	REFERENCE SECTION	✓	Initial
	300 HOUR INSPECTION (cont)			
	NOTE: Compliance with M250 CEB 72-3059, 72-3096, 72-3100, A-72-3134 (twin engine applications), and A-72-3135 (single engine applications) eliminates this inspection requirement.			
41	Inspect the thermocouple assembly (TOT/MGT).	77-20-01 PARA 2.B.		
	600 HOUR INSPECTION			
	The following inspection is required every 600 hours time since last inspection.			
1	Do the scavenge oil filter impending bypass function check as in Facet Service Bulletin No. 090589 (Ref. Rolls-Royce CSL 3116) for all aircraft equipped with an external scavenge filter system. Follow the Facet instructions and time intervals, or follow this recommended inspection interval each 600 hours.			
	2000 HOUR INSPECTION			
	The following inspection is required every 2000 hours time since last inspection.			
1	Fuel control filter inspection.	73-20-02, PARA 5.A.		
2	Fuel nozzle filter inspection (if applicable)	73-10-03, PARA 3.C.		
3	Remove and replace the low pressure fuel filter element. Before discarding filter, inspect for signs of contaminants. If any are found, inspect the entire fuel system and clean if necessary.	73-10-05, PARA 2.		
4	Inspect the combustion liner.	72-40-00, PARA 1.C.		
5	Examine the outer combustion case for cracks using Leak-Tek and/or Fluorescent Penetrant Inspection (FPI).	72-40-00, PARA 2.B.(2) (3), and (4).		
6	Inspect the compressor discharge air tubes.	72-40-00, PARA 4.C.		
7	Inspect the N_2 overspeed mounting dampers.	73-21-00, PARA 7.B.		

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Item	Inspection/Maintenance Action	REFERENCE SECTION	✓	Initial
	2000 HOUR INSPECTION			
8	Inspect the spur adapter gearshaft, compressor rotor splined adapter and associated impeller bore.	72-30-00, PARA 4.B.(2), 4.C. and 4.E.		
9	Inspect the turbine to compressor coupling, turbine splined adapter, power turbine inner shaft and turbine shaft-to-pinion gear coupling. Turbine to compressor coupling is part of the turbine assembly.	72-50-00, PARA 6.A. and 6.B.		
	NOTE: Not applicable for: Torquemeter gear part number 23084248 and subsequent Power take--off gear part number 23084249 and subsequent Pinion gear part number 23084247 and subsequent.			
10	Examine the power drivetrain gears. Disassembly of the gearbox is not necessary for this inspection.	CSL 3225		
	NOTES: Not applicable for: Torquemeter gear part number 23084248 and subsequent. Power take-off gear part number 23084249 and subsequent. Pinion gear part number 23084247 and subsequent.			
	NOTES: The following inspections are recommended whenever the turbine or compressor is removed in-between the required 2000 hour inspection. Anytime the compressor is removed from the engine, visually inspect the aft end of the spur adapter gearshaft for worn or damaged splines. Anytime the turbine is removed from the engine, visually inspect the splines on the following items, turbine-to-compressor coupling, turbine splined adapter, power turbine outer shaft and turbine shaft-to-pinion gear coupling for worn or damaged splines. If spline wear or damage is observed the appropriate maintenance action is required. (Refer to item 8 and 9 above).			
	Inspection intervals must not exceed 2000 hours.			

I certify that this aircraft has been inspected in accordance with Rolls-Royce M250-C30 Series 2000 Hour Engine Inspection and was determined to be in an airworthy condition.

Signature: _____ Cert #: _____

Registration No.: N_____ Aircraft Ending Total Time: _____

Aircraft S/N:_____ Date:_____ Engine Ending Total Time: _____

Engine S/N: _____ Engine Ending Total Cycles: _____

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